

Experimental Pulsatile Perturbations of Near Wall Thermal Transport

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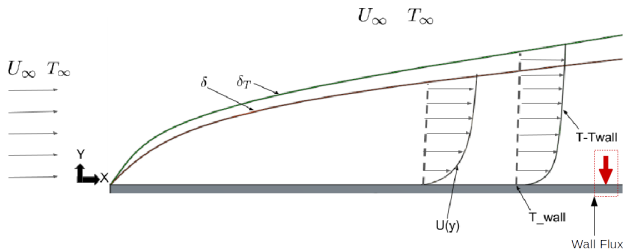
THIS WORK IS SUPPORTED BY AN NSF/DOE
PARTNERSHIP

AWARD'S: 1258702, 1258594, 1258697



Thermal Boundary Layers 101

-2 field variables to study, Velocity and Temperature



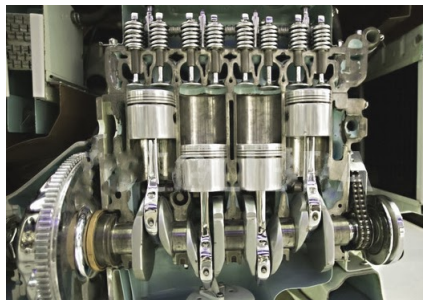
- ▶ U_∞ = Free-stream velocity
- ▶ $U(y)$ = Stream-wise velocity profile
- ▶ δ = Boundary layer thickness
- ▶ $Pr = .707 = (\text{viscous diff.}) / (\text{thermal diff.})$
- ▶ T_∞ = Free-stream temperature
- ▶ T_{wall} = Wall temperature
- ▶ δ_T = Thermal boundary layer thickness

Engineering Systems

-Design of engineering systems relies on accurate models of real physics



<http://science.howstuffworks.com>



<http://me-mechanicalengineering.com>

Important Parameters:

Wall shear stress $\rightarrow \tau_{wall}$

Wall heat flux $\rightarrow q''_{wall}$

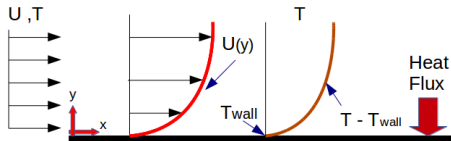
Complex Thermal Boundary Layers

Complex Boundary Layers
brought about from;

- ▶ Induced pressure gradients
- ▶ Temperature gradients
- ▶ Separation
- ▶ Dynamic walls
- ▶ Unsteady flow

In many engineering applications
one or several of these effects are
important

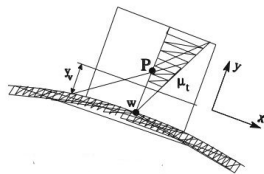
Equilibrium Boundary Layer



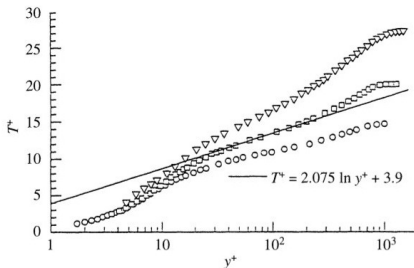
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Modeling Non-Equilibrium Flows

- ▶ Computationally expensive to solve near wall dynamics
- ▶ Near wall dynamics extrapolated from log profile
- ▶ Reasonable estimate for equilibrium wall flows
- ▶ Fail when applied non-equilibrium boundary layers



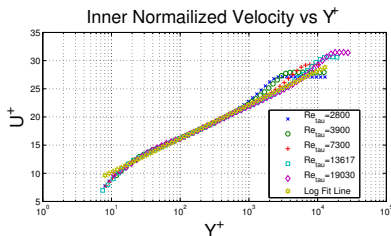
P. Bradshaw and G. P. Huang



Bradshaw et al., *The law of the wall in turbulent flow*

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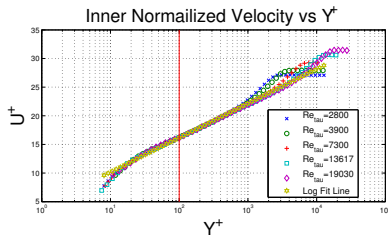


Hutchins et. al (2009)

- ▶ For $Re = 9.7 \times 10^6$ ($\diamond$)
- ▶ $\delta = .302m$
- ▶ Scaling $y^+ = \frac{yu_{\tau}}{\nu}$, $u^+ = \frac{u}{u_{\tau}}$
- ▶ Viscous length scale = 16 microns

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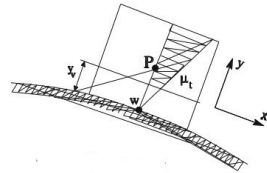
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- ▶ Viscous length scale = 16 microns
- ▶ $y^+ = 100 \rightarrow y = 1.6mm$

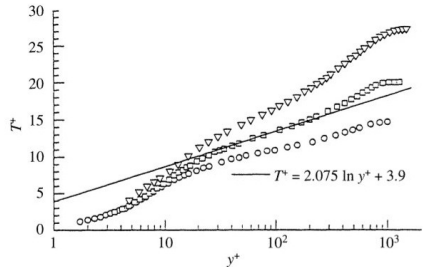
Modeling Non-Equilibrium Flows

Question

How does time varying boundary conditions of U_∞ change the momentum and thermal boundary layer profiles?



P. Bradshaw and G. P. Huang

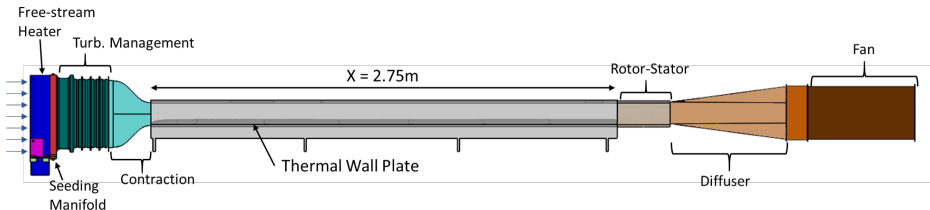


Bradshaw et al., *The law of the wall in turbulent flow*

Experimental Facility

Non-Equilibrium And Thermal boundary layer Tunnel (N.E.A.T.)

- Length=2.75m
 - Turbulent Management section
 - VFD Controlled Motor
 - Bank of Resistive Heaters
 - Thermal Wall Plate
 - Rotor-Stator Mechanism
- > Development Length
 —————> Free Stream Turbulence
 —————> U_{∞}
 —————> T_{∞}
 —————> T_{wall}
 —————> $\frac{\partial U_{\infty}}{\partial t}$

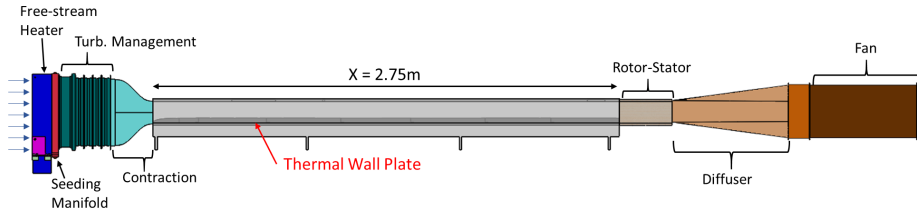


UNH Thermal Boundary Layer Wind Tunnel

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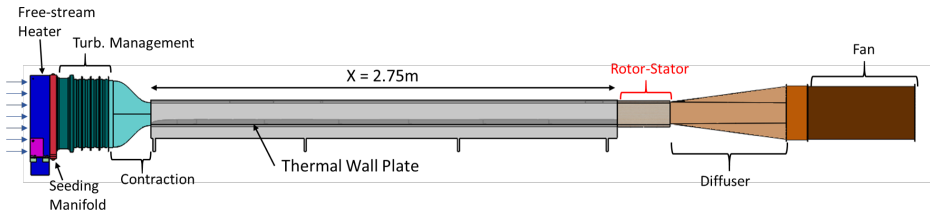


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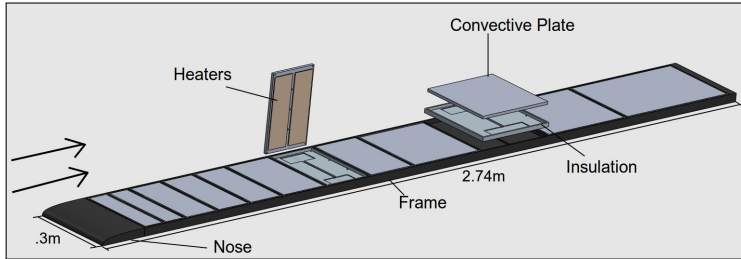
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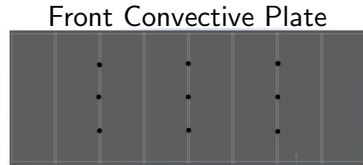


UNH Thermal Boundary Layer Wind Tunnel

Sectioned Thermal Wall Temperature Plate



- ▶ Size=0.3m x 2.74m
- ▶ Sectioned design*
- ▶ Independently heated/controlled*
- ▶ Individually Insulated

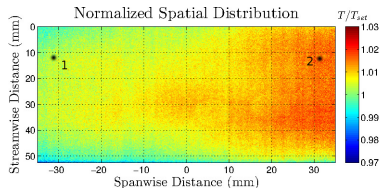


* Blackwell, B. F., *The turbulent boundary layer on a porous plate*

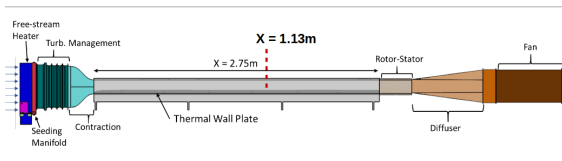
D. Biles et al., UNH, SHTC 2017

Validation of Wall Plate Design

- ▶ Control set Temperature to $\pm 0.1^\circ\text{C}$
- ▶ Produce spatially uniform temperature to $\pm 2\%$ in equilibrium flow
- ▶ Develop 2D equilibrium thermal boundary layer

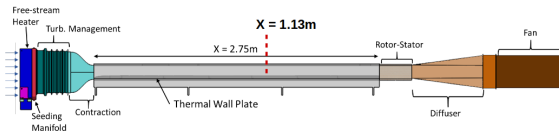
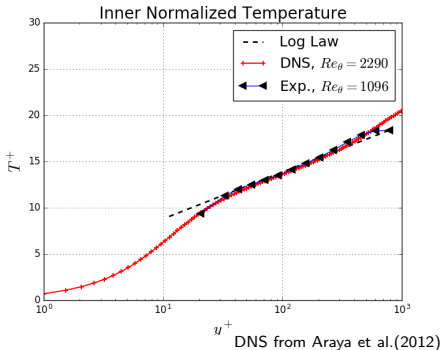


Thermocouple #1 = 50°C
Thermocouple #2 = 50.01°C



Validation of Wall Plate Design

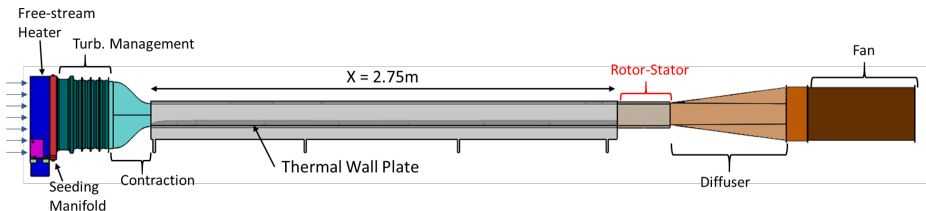
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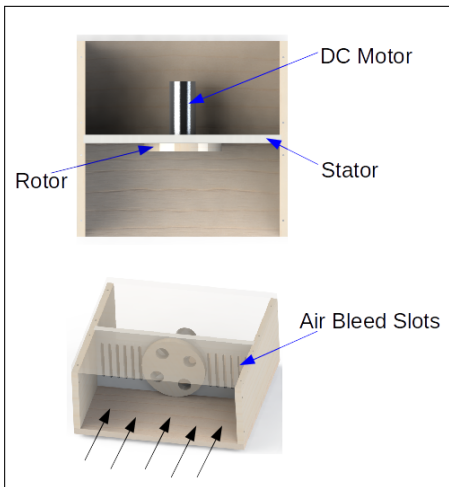
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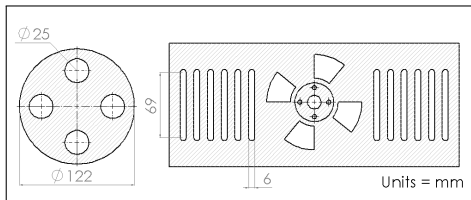


UNH Thermal Boundary Layer Wind Tunnel

Rotor-Stator Design



- ▶ Slotted Rotor-Stator design
- ▶ Rotor outer diameter = channel height
- ▶ Adjustable number of air-bleed slots
- ▶ Rotor speed adjustable from 1Hz to 85Hz

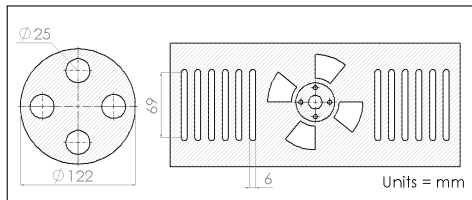
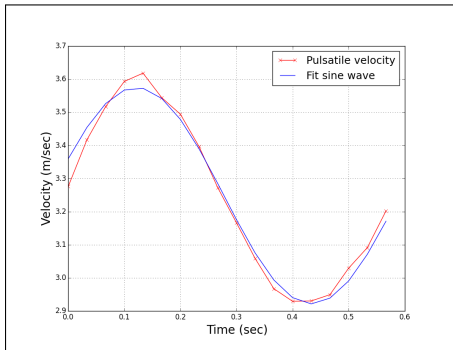


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¹K. Al-Asmi and I.P. Castro, *Production of oscillatory flow in wind tunnels*

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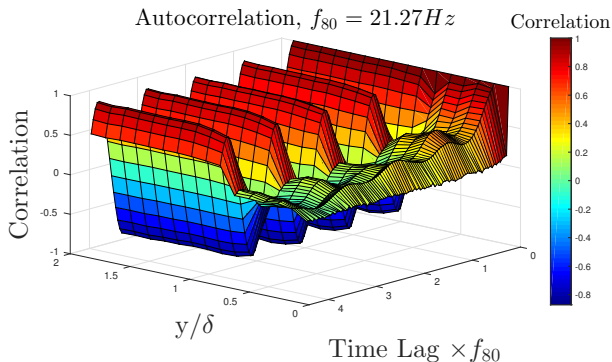
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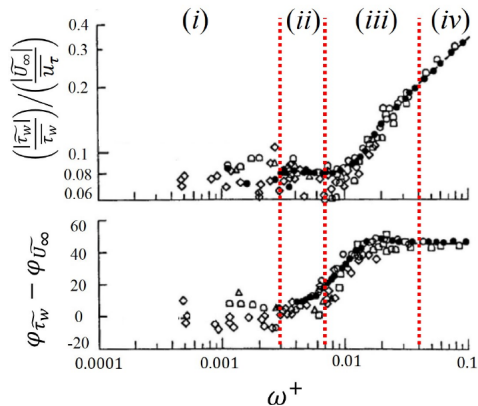
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Rotor-Stator Testing: Auto-correlation



- ▶ Induced pulsatile flow in free stream
- ▶ Correlation drops off within boundary layer
- ▶ Correlation peaks still visible in boundary layer

Non-Equilibrium Classification

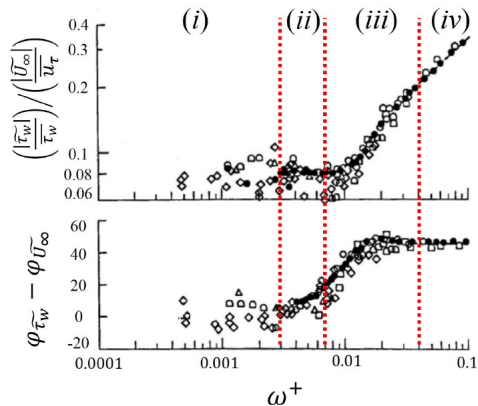


- ▶ (i) - Quasi-steady region
- ▶ (ii) - Low frequency region
- ▶ (iii) - Intermediate frequency region
- ▶ (iv) - High frequency region

Markers from Brereton and Mankbadi (1995)

Modulation Amplitude (top), Phase Shift
(bottom) of perturbation wall shear stress

Non-Equilibrium Classification

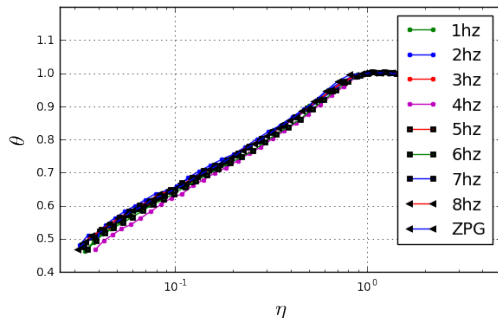


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Pulsatile Mean and Integral Parameters



▶ $\delta_T / TC_{diam} = 28$

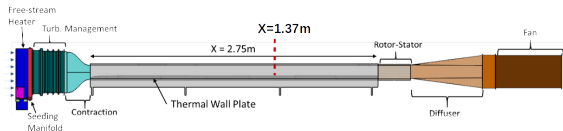
▶ $\theta = \frac{T_{wall} - T(y)}{T_{wall} - T_{\infty}}$

▶ $\eta = y / \delta_T$

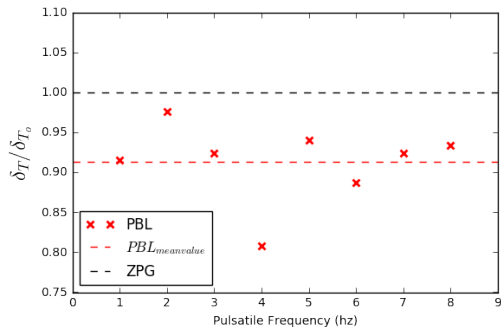
▶ $Re = 2.6 \times 10^5$

▶ $T_{wall} = 40^\circ C$

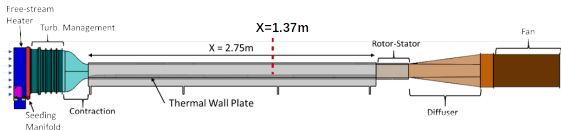
▶ Frequency =
1Hz $\rightarrow$ 8Hz



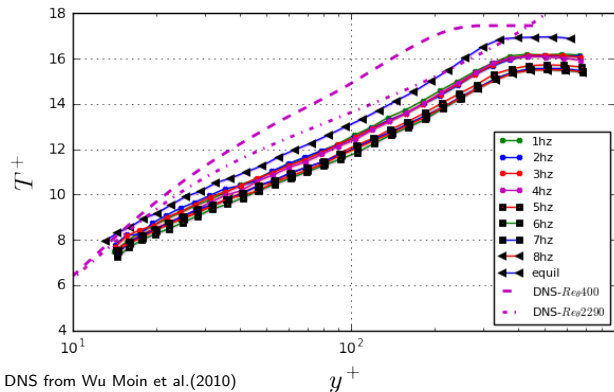
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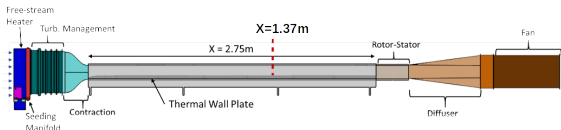


Inner Normalized Profiles

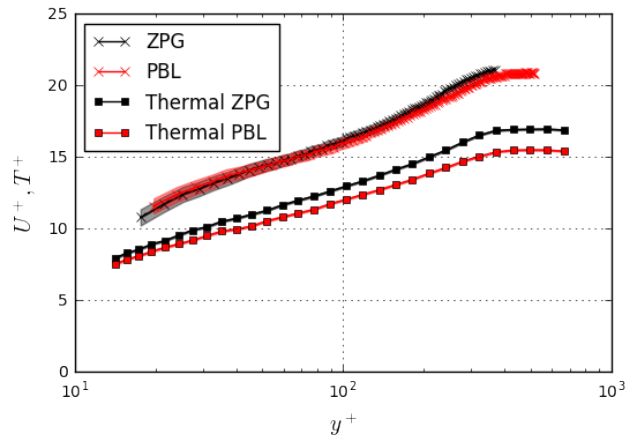


ZPG

- ▶ $T^+ = \theta \frac{U_{\infty}}{u_{\tau}} Pr^{-2/3}$
- ▶ $y^+ = \frac{yu_{\tau}}{\nu}$
- ▶ $Re = 2.6 \times 10^5$
- ▶ $T_{wall} = 40^{\circ}C$
- ▶ Frequency = 1Hz $\rightarrow$ 8Hz



ZPG vs Pulsatile Flow



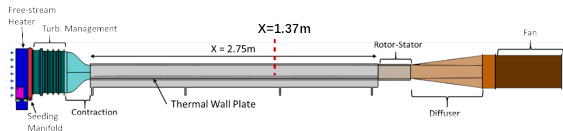
ZPG

► $Re_\theta = 1139$

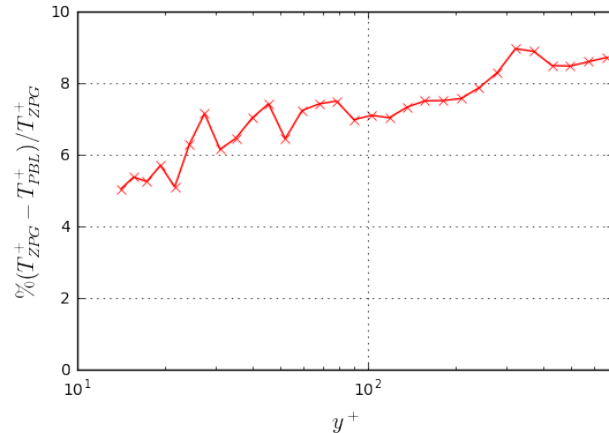
Pulsatile

► $Re_\theta = 874$

► Freq = 6Hz



ZPG vs Pulsatile Flow



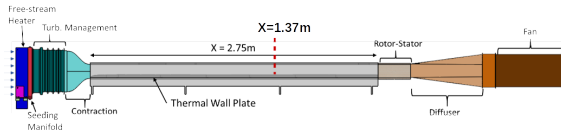
ZPG

► $Re_\theta = 1139$

Pulsatile

► $Re_\theta = 874$

► Freq = 6Hz



Conclusions

- ▶ Pulsatile forcing of outer velocity boundary condition results in pressure gradient like influence on temperature field
- ▶ Since the effect of a favorable pressure gradient is most evident in the inner-normalized profiles it is possible that pulsatile forcing modifies the wall heat flux or causes a phase lag between the wall shear stress and the wall heat flux.
- ▶ Both results suggest a breakdown of Reynolds analogy in pulsatile flow in the intermediate non-equilibrium region. Perhaps not surprising since Bradshaw showed that Reynolds analogy breaks down in adverse and favorable pressure gradient flows

Future Work

- ▶ Phase averaged wall heat flux
- ▶ Coupled temperature and velocity measurements to investigate $v' T'$ correlation

Thank you for Your Attention!

